

DESIGN OF A 207 FT. SPAN SPANDREL-
BRACED TWO-HINGED ARCH

BY

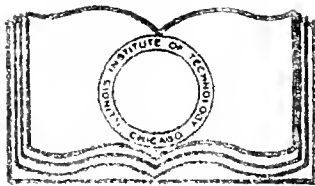
R. L. STEVENS and W. TRINKAUS

Armour Institute of Technology

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Design of a 207 ft. span
spandrel braced two-hinged

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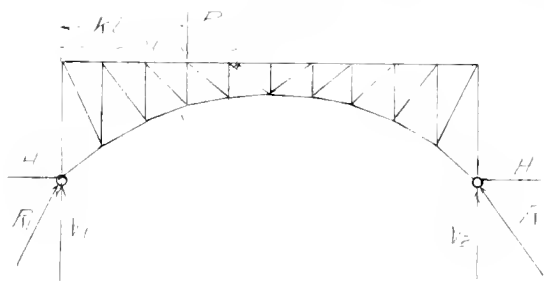
Handwritten signatures and notes:
J. M. Pagnon
J. M. Pagnon
J. M. Pagnon
J. M. Pagnon

In arches of three hinges, the horizontal thrusts are all equal. The hinges provide three points at which the reaction of the arch section to bending is zero. Since there are three components of the reactions to be found & the vertical reaction at each end and the horizontal thrust at each, write a moment for the bending moment in the arch in terms of the load and reaction, at each hinge, set the value equal to zero, since at these points the arch can not bend; stress, and find the value for the three reactions & the thrusts.

Standard-bored over the center of Florida for the purpose of
 such a way that it will be completed. That it is not possible
 so to eliminate the extent of forested area.

1. It is required to determine the reaction forces at the supports of a beam of length l and weight G supported by two supports. The beam is supported by a hinge support at the left end and a roller support at the right end. The weight of the beam acts vertically downwards at the center of gravity, which is at a distance of $l/2$ from the left end. The reaction forces at the supports are denoted by R_1 and R_2 respectively. The reaction force at the hinge support R_1 has both horizontal and vertical components, while the reaction force at the roller support R_2 has only a vertical component. The horizontal component of R_1 is denoted by H and the vertical component by V_1 . The reaction force at the roller support R_2 is denoted by V_2 . The weight of the beam G acts at the center of gravity, which is at a distance of $l/2$ from the left end. The reaction forces at the supports are determined by the equilibrium conditions. The sum of the horizontal forces must be zero, the sum of the vertical forces must be zero, and the sum of the moments about any point must be zero. The reaction forces at the supports are $R_1 = H$ and $R_2 = V_2$.

Deflection of the beam is given by:



The reaction forces at the supports are determined by the equilibrium conditions. The sum of the horizontal forces must be zero, the sum of the vertical forces must be zero, and the sum of the moments about any point must be zero. The reaction forces at the supports are $R_1 = H$ and $R_2 = V_2$.

3. The reaction forces at the supports of a beam of length l and weight G supported by two supports. The beam is supported by a hinge support at the left end and a roller support at the right end. The weight of the beam acts vertically downwards at the center of gravity, which is at a distance of $l/2$ from the left end. The reaction forces at the supports are denoted by R_1 and R_2 respectively. The reaction force at the hinge support R_1 has both horizontal and vertical components, while the reaction force at the roller support R_2 has only a vertical component. The horizontal component of R_1 is denoted by H and the vertical component by V_1 . The reaction force at the roller support R_2 is denoted by V_2 . The weight of the beam G acts at the center of gravity, which is at a distance of $l/2$ from the left end. The reaction forces at the supports are determined by the equilibrium conditions. The sum of the horizontal forces must be zero, the sum of the vertical forces must be zero, and the sum of the moments about any point must be zero. The reaction forces at the supports are $R_1 = H$ and $R_2 = V_2$.

By taking moments of the external forces about either A or B , it is seen that

$$V_1 = P(l/k) \quad \text{and} \quad V_2 = Pk$$

which is the same as the reactions on a beam of equal span under the same load.

The value of H is the only part of the reactions now undetermined and in the following its value is deduced.

Formulae for Horizontal Thrust.

The following notation will be used:

P = simple vertical load on arch

k/l = distance of load from left abutment

R_1 = reaction at left hinge

R_2 = " " right " "

V_1 = vertical component of left reaction

V_2 = " " " right " "

H = horizontal component of reactions

S'_n = stress which would exist in any member from the vertical components only, of the reactions

T_n = stress which would exist in any member from a horizontal reaction of value H

S_n = actual stress in any member from load P

A_n = sectional area of any member

L_n = length of any member

E = modulus of elasticity of steel

δ_n = deformation of any member due to load P

Δ = horizontal deflection of structure which would take place under P if one end were free to move laterally.

∴ "The partial derivatives of the Helmholtz free energy with respect to the statistically independent parameters of the system are equal to the corresponding thermodynamic quantities."

This implies the following relations for the derivatives of the Helmholtz free energy with respect to the parameters:

The work done by the system is equal to the negative of the change in Helmholtz free energy:

$$dW = -\frac{1}{2} S \delta$$

For the Helmholtz free energy, the work done is equal to the negative of the change in Helmholtz free energy:

$$\delta = \frac{S \delta L}{A E}$$

$$\therefore dW = \frac{S^2 \delta L}{2 A E}$$

The total work done by the system is equal to the negative of the change in Helmholtz free energy:

$$W = \sum \frac{S^2 L}{2 A E}$$

For

$$S = S' + HT$$

$$\therefore W = \sum \frac{(S' + HT)^2 L}{2 A E}$$

For constant temperature, the work done is equal to the negative of the change in Helmholtz free energy:

$$-\Delta F = \frac{dW}{dH}$$

where ΔF is the change in Helmholtz free energy, and H is the temperature.

∴ $\frac{dW}{dH} = -\Delta F$ is the negative of the change in Helmholtz free energy with respect to the temperature.

But

$$W = \sum \frac{(S'^2 + 2S'TH + T^2 H^2) L}{2 A E}$$

$$\therefore \frac{dW}{dH} = \sum \frac{(2S'T + 2TH^2) L}{2 A E}$$

$$\therefore -\Delta F = \sum \frac{S'T L}{A E} + H = \frac{TS L}{A E}$$

$$H = -\frac{\sum \frac{S'T L}{A E} + \Delta F}{\sum \frac{T L}{A E}}$$

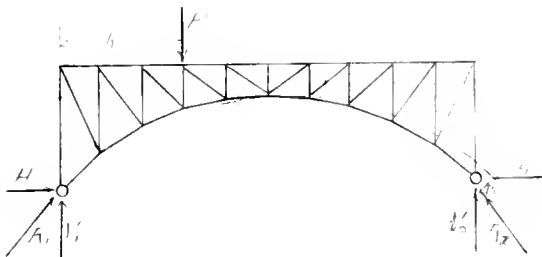
If T is constant, then

the denominator is constant, and $H = -\Delta F / T$

$$\therefore H = -\frac{\sum \frac{S'T L}{A E}}{\sum \frac{T L}{A E}}$$

A truss is shown

with a horizontal load



to the right.

1. Find P and H .

2. Find the reaction at the

right support.

3.

If the load P

is just sufficient to

make the truss

just to the right of the

horizontal force at

the hinge is the same

If we divide the

the first case, the

the second case, the

the third case, the

the fourth case, the

the fifth case, the

the sixth case, the

the seventh case, the

the eighth case, the

stress is any longer caused by Q and δ the corresponding deformation in the member.

The external work done by Q in causing the deformation Δ' is

$$W = \frac{1}{2} Q \Delta'$$

and the internal work in any member is

$$dW = \frac{1}{2} T \delta$$

The total internal work of deforming the truss is the sum of the work done on each member, or

$$W = \frac{1}{2} \sum T \delta$$

Since the total internal work is equal to the external work

$$Q \Delta' = \sum T \delta$$

P is the load on the truss, suspended from the joint at H , S the stresses, S' the stresses in the bars, λ the coefficient of expansion, Δ the

movement of hinge due to P . Then $\Delta = \frac{1}{2} H \Delta$ and the

$$\text{work} = \frac{1}{2} \sum S \lambda$$

And

$$\lambda = \frac{S' L}{A E}$$

The deformation Δ' is the sum of the deformations of the bars, or the sum of δ , or $\frac{S' L}{A E}$, or $\frac{S' L}{A E} \sum \lambda$.

Let Δ'_1 be the deformation of the deflection of the truss due to Q to the deformation at any point.

$$\therefore \frac{1}{2} Q \Delta'_1 = \frac{1}{2} T \delta$$

$$\therefore \Delta'_1 = \frac{S' L}{A E} \sum \lambda = \frac{S' L}{A E} \sum \lambda$$

also, let Δ'_2 be the deformation of the truss due to P only, due to the load P at the joint at H and the deflection at H .

$$\therefore \frac{1}{2} H \Delta'_2 = \frac{1}{2} S \lambda$$

$$\therefore \Delta'_2 = \frac{S' L}{H} \sum \lambda$$

Let for any element, $\frac{S'}{H} = \frac{T}{Q}$ since the force H is the same as the member divided by the force, strain at the surface, T , which causes the stress. (Stress is the force divided directly, proportionally to the intensity of the forces causing the.)

Substituting for $\frac{S'}{H}$ its equal, $\frac{T}{Q}$,

$$\Delta_1 = \frac{T}{Q} \Delta = \frac{T}{Q} \frac{S' L}{A E}$$

This is the portion of the deflection of the hinge due to the force H at one corner. The total deflection of the section of these portions is the horizontal distance, Δ ,

$$\Delta = \sum \Delta_1 = \sum \frac{T}{Q} \frac{S' L}{A E}$$

Since the actual value of Q is invariable, let its value be unity, so that

$$\Delta = \sum \frac{S' T L}{A E}$$

This gives the deflection of the rigid member. When a load P is

We have now to derive proper value of Δ from the stress H in pushing and pulling the hinge back to its original position.

Let H be the force applied equal in magnitude to the Δ as defined in the hinge. The stress δ is the force HT as deformation, δ .

$$\text{External work} = \frac{1}{2} H \Delta = \text{Internal work} = \frac{1}{2} H T \Delta$$

$$\therefore \Delta = \sum T \delta$$

But

$$\begin{aligned} \delta &= \frac{HT}{A E} \\ \Delta &= \sum \frac{T^2 H L}{A E} \\ &= H \sum \frac{T^2 L}{A E} \end{aligned}$$

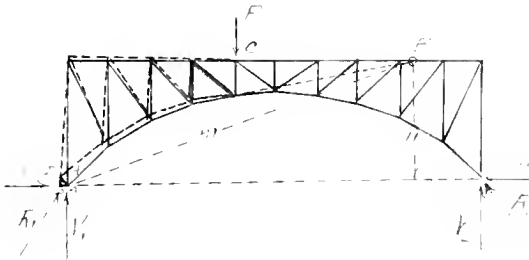
Equating the two values of Δ obtained,

$$\sum_{AE} \frac{STL}{AE} = H \sum_{AE} \frac{1}{AE}$$

$$\therefore H = \frac{\sum_{AE} \frac{STL}{AE}}{\sum_{AE} \frac{1}{AE}}$$

At the crown of the arch, $\theta = 0$, $\therefore H = 0$

Ex. 1. Find the horizontal displacement of the crown of the arch.



Let Δ be the horizontal displacement of the crown. The horizontal displacement of the crown is the horizontal distance between the vertical line through the crown of the undeformed arch and the vertical line through the crown of the deformed arch.

Let Δ be the horizontal displacement of the crown.

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$$\therefore \frac{d}{m} = \frac{\lambda}{V}$$

Let Δ be the horizontal displacement of the crown.

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Let Δ be the horizontal displacement of the crown.

and

$$\therefore \frac{\Delta_1}{y} = \frac{d}{m} = \frac{1}{10}$$

$$\therefore \Delta_1 = \frac{y}{10}$$

L. HT

H. S'

and

and

S

and

$$\therefore S = S' + HT$$

$$\Delta = \frac{S L}{A E}$$

$$= \frac{S' L}{A E} + \frac{HT}{A E}$$

$$\therefore \Delta_1 = \frac{y}{10} = \frac{S' L}{A E} + \frac{HT}{A E}$$

$$= \frac{S' L}{A E} + \frac{HT}{A E}$$

$$\frac{y}{10} = T$$

$$\therefore \Delta_1 = \frac{S' L}{A E} + H \frac{T}{A E}$$

and

$$\Delta = \Delta_1 = \frac{S' L}{A E} + H \frac{T}{A E}$$

and

$$\therefore \Delta = \frac{S' L}{A E} + H \frac{T}{A E}$$

$$= \frac{S' L}{A E}$$

$$= \frac{S' L}{A E}$$

and

and

and

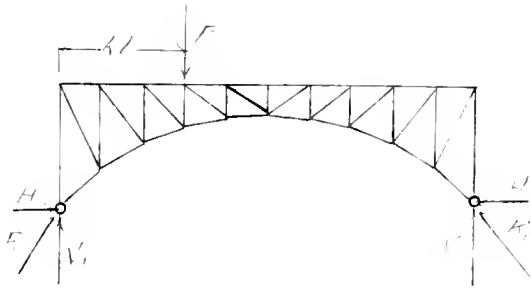
and

$$\frac{S' L}{A E}$$

$$\frac{S' L}{A E}$$

$$\frac{S' L}{A E}$$

$$H = \frac{\sum \frac{S' T L}{A}}{\sum \frac{T L}{A}}$$



$$S' = V_2 \frac{L}{V}$$

$$S' = (V_1 + V_2) \frac{L}{V}$$

$$= F \frac{L}{V}$$

$$S' = \frac{L}{V} (V_1 u - P(u - H))$$

$$V_1 = F(1 - \frac{H}{L})$$

$$= F - \frac{V_2}{F}$$

$$\therefore S' = \frac{L}{V} (V_1 u - F u + F H)$$

$$S' = \frac{V_2 L}{V}$$

For the stresses in both members,

$$S' = (V_1 + V_2) \frac{U}{V} + \frac{1}{V} (P_l - P_u - V_1 l) \\ - P \frac{1}{V} - \frac{P_l}{V} + \frac{P k l}{V} \\ - P k \frac{1}{V}$$

Since the truss is symmetrical, T , the stress due to horizontal thrust is the same in the symmetrical members.

Substituting for S' in the formula for H

$$H = \frac{\sum_0^{k l} P \frac{U}{V} \frac{T l}{A} + \sum_0^{k l} P \frac{k l}{V} \frac{T l}{A}}{2 \sum_0^{k l} \frac{T^2 l}{A}}$$

the summation covering the members to the left of the load P .

$$\therefore H = P \frac{\sum_0^{k l} \frac{U}{V} \frac{T l}{A} + \sum_0^{k l} \frac{k l}{V} \frac{T l}{A}}{2 \sum_0^{k l} \frac{T^2 l}{A}}$$

Let n be the number of the panels to the left of the load P is acting. Let p be the panel length.

Then the distance of the load from the left end,

$$k l = n p. \\ \therefore H = P \frac{\sum_0^{n p} \frac{U}{V} \frac{T l}{A} + n \sum \frac{k}{V} \frac{T l}{A}}{2 \sum \frac{T^2 l}{A}}$$

The stress in any member,

$$S = S' + H \cdot F$$

Let N be the number of panels in the truss.

$$\therefore l = N v \\ V_1 = P \left(1 - \frac{k l}{l} \right) = P \frac{N p - n p}{N p} \\ = P \frac{N - n}{N} \\ \therefore S' = \frac{V_1 U}{V} = \frac{N - n}{N} \frac{U}{V}$$

For members to the left of P

$$S' = V_0 \frac{1}{V} - P \left(\frac{U - \eta P}{V} \right) \\
= P \frac{N - \eta}{N} \frac{1}{V} - P \frac{U - \eta P}{V} \\
P \left(\frac{N - \eta}{N} \frac{1}{V} - \frac{U - \eta P}{V} \right)$$

for members between P and the middle of the truss.

Substituting these values of S' and H in the formula for S

$$S = P \left[\frac{N - \eta}{N} \frac{U}{V} - \frac{U - \eta P}{V} + \left(\frac{\sum_c \frac{U T L}{V A} + \eta \frac{\sum_c \frac{P T L}{V A}}{\frac{E_c - E T^2 L}{A}} \right) / \right]$$

which gives the stress in any member due to load P at any point η . The second term may be dropped, however, except for members between P and the middle of the truss.

The Design.

The bridge selected to be designed according to the model is of the same general dimensions as one designed and built in 1902 by the Chicago, Milwaukee and St. Paul Railroad at Iron Point, Michigan, as a three-chord steel arch. The span is 200' and the depth 82'. It is a single track deck structure with trusses spaced 22' apart and a floor beam.

The bridge crosses the Menominee River and at that point the banks consist of solid granite so that the situation is ideally an arch span.

For this design the crown depth was assumed as 8 feet, the curve of the lower chord as a parabola, and the span was divided into ten panels of 20' in length. The same loadings, unit stresses and specifications were used as with the two designs given in a measure as a basis for comparison of the two classes of arches. The outline of one-half of the truss with the lengths and radii of arcs of the members is given in Plate 2.

The live load is that known as Cooper's "Class E-10 Road" except that the uniform load follows the two locomotives and is assumed as 7000 pounds per foot of track instead of 6000 pounds to allow for the excessive weight of ore trains. The intensity of the uniform load was so great that it was used instead of locomotive concentrations in finding the stresses in the trusses. For the floor system the moments and shears were greater for the concentrated loads than for the uniform load. The length of the locomotive (without tender) wheel base being nearly equal to a panel length, the difference between its weight and that of an

equal length of uniform load was taken as an "excess" load and was applied to two such alternate panel points as would produce the maximum stress in each member. The details of the loading are given in table 1.

The fact that the bridge is anchored only at the abutment hinges, while the correct position of the trusses and connections and the live loads exposed to wind action are at a considerable distance above the anchorage, gives rise to large overturning moments which act to produce loads on the truss, acting downward on the leeward side and upward on the windward side.

The distribution of the wind stresses among the various systems of bracing is undetermined in this kind of a structure, but by making the upper lateral bracing of nominal dimensions we have considered that the loads applied on the upper panel points are carried down through the cross bracing to the lower panel points and from there through the lower system of lateral bracing to the abutments. This is probably the most direct way of transferring the wind loads to the abutments and was taken as the most probable.

The wind on the train is assumed to act 3 feet above the middle of the upper chord and is treated as live load. This live wind load and that applied to the upper chord produces an overturning moment about the connection of the lower chord panel points. Since the lower chord panel points are not in the same horizontal plane, the load at each panel point produces an overturning moment about the next panel point toward the abutment. The loads are given in Table 1. In addition to the vertical

loads on the truss . . . overturning, the horizontal wind loads acting on the lower chord and transferred to it by the sway frames produce stresses in the lower lateral system. A graphical determination of the stresses in the lower lateral system is given on Plate II and in Table II is given the composition of the lower lateral system.

The design of the intermediate sway bracing is given on Plate III. The stresses in the end sway bracing and in the lattice beam are given, graphically, in Plate IV and in Table III. The design of the floor system is given on Plate V.

As a preliminary to the computation of the stresses in the truss of constants for the members of the truss was computed. These are independent of the loads on the truss and are given in Table V.

The stresses in each member due to loads of unit magnitude at various panel points, considering only the effect of the vertical reactions, were next computed. This corresponds to the terms

$$\frac{N-n}{N} \frac{u}{V} - \frac{u-np}{V}$$

The former is given in Table VI and the latter in Table VII for the members and loads to which it applies. They are combined in Table VIII.

The term $\sum \frac{1}{V} TL + n \sum \frac{P}{V} L$

was next computed. Since in this case trial we have nothing to determine the areas of the sections, they are assumed for this purpose to be all equal so that the term cancels out of the expression for H . The values for this expression are given in Table IX, the quantities for each class of members above the

heavy lines being computed from the first term and those below from the second. Since for panel loads beyond 4 there are no members to the right of the load, then for all succeeding panel loads only the first term applies. The summations are obtained by adding all the quantities for one panel load, all the S' 's being minus as S' 's of opposite sign to T . The summations for each panel load are divided by the quantity,

$$L \sum_0^{2L} T^2 L$$

from Table V, and the result is the value of

$$\sum \frac{H}{V} T L + \sum H \frac{P}{V} T L \\ = \sum T^2 L$$

for each unit panel load in the expression for H .

In Table X these values have been multiplied by the value of T for each member.

In Table XI these stresses from horizontal reactions are combined with those from vertical reactions in Table VIII and the results are the actual stresses in the members from unit panel loads.

Since for dead load, all panel loads have to be considered as acting at all times and as the same load is concentrated at the corresponding panel points from the two ends, some labor is saved by combining the corresponding unit stresses before multiplying them by the panel loads. These are given in Table XII. Since the live and "excess" panel loads are all equal, the unit stresses which will produce the largest values, plus and minus, are combined in Table XII. In Table XIII are given the stresses due to dead panel loads, being the values in Table XII multiplied by the panel loads. These are combined and are given in the

column for Dead Load in Table XV. In Table XIV are given stresses due to wind loads, and under wind load in Table XV the sums of the stresses of the same sign are given. For the live loads the maximum possible stresses on each panel point will take place when every panel point which gives stresses of one kind is loaded and when all panel points which give stresses of the opposite kind are unloaded. In the columns for live load and dead load the quantities in the corresponding columns for wind stresses in Table XIV are multiplied by the panel loads. The specified unit stresses provided for the structural area for live load stresses shall be twice that provided for wind dead load stresses. For this reason the specified unit stresses for the stresses in the last two columns are half the specified live stresses and half the dead load stresses, and the live and dead stresses used in designing. In Table XVI is given a preliminary design of the members for which areas are obtained. The areas used in the second trial are given. At this point it was thought best to provide a factor of safety of 1.5 in the design. In this preliminary, the stresses should be increased 50%. The results of the second trial are given. It is to be noted that since the total stresses were increased 50% the areas were increased 50%. The areas were increased 50% and the stresses were increased 50%.

From the areas figured, a weight of dead load was ^{computed} and was used instead of the original dead load in the design. The weights were figured from the areas of the members and the

to provide for the additional weight of components in there of 40 was 16. The dead load, in pound weight is given in Table XVII.

In Table VIII are given the computations of several constants in the formulas, including the value of the stress.

The temperature stresses were figured from modification of the formula for horizontal thrust. In the derivation of the formula for horizontal thrust by the author it is assumed that

$$H = \frac{E A \Delta T \alpha}{L}$$

where ΔT is the change in T .

If for ΔT is put the value of the change of T which would take place from the change in length of members from change in temperature if the ends were unrestricted, the horizontal thrust of H would be a thrust upon the abutments caused by the tendency to change of span.

$$H = \frac{E A \Delta T \alpha}{L}$$

where α is the coefficient of linear expansion per degree Fahrenheit for steel, ΔT the change in temperature above or below the normal and L the span. The term

$$\frac{E A \Delta T \alpha}{L}$$

is zero since for horizontal span only ΔT is zero.

The stress in any member due to temperature will then be the horizontal thrust due to that of a member multiplied

The following values for the quantities were taken.

$$E = 30000 \text{ mps}$$

$$G = 1.0000000$$

$$L = 10 \text{ m}$$

$$t = 100 \text{ s}$$

$$S_f = \frac{30000 \times 1.0000000 \times 10}{\sum \frac{T^2}{A}} = \frac{300000}{\sum \frac{T^2}{A}}$$

From Table XVIII

$$\sum \frac{T^2}{A} = 24.315$$

so that $S_f = 12345.67$

The values of S_f for each member is given in the last column in Table XVIII.

Tables XIX to XXVI are similar to those previously mentioned and are self explanatory.

In Table XXVII is given the detailed designs of the sections as finally adopted and of Plate VI a general outline of the arch.

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